Problem description

To verify that you understand the somewhat tricky input features of a space frame (general beam bending, torsion, and axial load) it is wise to start with a published solution. An example from [1] is a three member spatial "Z" shaped space frame that is fixed at both ends, and loaded with a vertical point load at one of the two free nodes. That simple system is illustrated in Figure 1. The top portion of the figure presents the desired orientations of the (assumed) rectangular cross-section of each of the three members. The bottom portion displays the usual line drawing representation used by most finite element systems for line elements. The data to be given below were simply taken from reference [1].

The beam is assumed to have a rectangular cross-section. Thus, the only tricky part of the study is to assure that the principle inertia axes of each rectangular beam section are properly orientated in 3D space. The desired member placement is seen in the top of \$ Figure while the usual line element format is in the bottom of that figure. The single line format can let you overlook a beam orientation error and possibly lead to seriously incorrect results.

Each beam has a local orthogonal coordinate system associated with it. The local x-axis runs from the first node to the second. It is used to compute the local torsional inertia, IXX. The local cross-section bending inertias, IYY and IZZ, are input relative to the local y- and z-axes, respectively. This is not a problem when the beam has a circular cross-section since IYY = IZZ for any choice of the local z-axis. That is not true for non-circular sections, so the local z-axis must be carefully established for each member (then the y-axis is computed by the "right hand rule"). The local z-axis can be defined in two ways in ANSYS. Here the method of using a third node will be employed. The three nodes define the local x-z plane, and therefore the cross-sectional axis used by you to compute IZZ. Knowing that, the IYY value is calculated with respect to the remaining orthogonal cross-section axis.

When selecting candidates for use as the third node it is usually preferred to pick an existing structure node. On some occasions you have to introduce a (full restrained) non-structural node simply to use as a geometric reference point. Ideally you could find structural nodes that make it easy to keep IYY and IZZ constant for a group of elements. Consider the first member in this system. It happens to go from node 1 to node 2. Try to pick a third node (from nodes 3, 4, or 5) that will allow the same IYY and IZZ values to be used in other elements. In Figure you can see that the plane formed by nodes 1-2-3 allow the first two members to share the same maximum and minimum local inertia axes, but the use of structural node 4 or geometry node 5 do not. Therefore, pick node 3 to define the local z-axis for member 1, as seen in Figure . Similarly, in Figure , the second member, along line 2-3 Is best, or more consistently, oriented by picking node 1 as its third node. For the last member, you have to make a choice between keeping the same IYY and IZZ values (i.e., same set of real constants) by creating node 5 as a non-structural node, or picking structure node 2 and reversing the IYY and IZZ values. In other words, picking node 2, which is done here, will

Smith Griffths space frame Ex. 4.4b الارد والي والار والعاصين والمانية والمانية ومواجع المراجع والمراجع والمراجع المراجع والمراجع والمراجع والمراجع والمراجع -----3 2 1 2 Smith Griffths space frame Ex. 4.4b

require a second set of real constants, and remembering when to activate that set. You will be reminded of these choices when the actual data are input.

Figure 1 A three member space frame with a vertical load

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Figure 2 Options fo rmembers local x-z planes (triangles)



Figure 3 Insert local axes for the first member (plane 1-2-3)



Figure 4 Choice for second member local axis (plane 2-3-1)

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Start ANSYS

ANSYS is short for **AN**alysis **SYS**tem. That pretty much sums up its capabilities. Most areas of engineering calculation are included: stress, thermal, fluids, dynamics, vibrations, frequency analysis, acoustics, electromagnetism, optimization, etc. are present and usually have non-linear abilities as well. ANSYS was first released in 1971 and has generally been one of the most popular finite element systems since then, worldwide. Since it has a huge list of abilities its menu system can get relatively long. That is also true because it (and all FEA systems at that time) was run in a "batch mode" on the largest available computers. In other words, you could, and still can, execute ANSYS through an input text file without using a Graphical User Interface (GUI). Once you are fully experienced with the code that can be a real time saver. Every GUI session is saved in that format so that you can simply edit it with a text editor and make small changes for a similar problem.

Begin ANSYS with **Start→All Programs→ANSYS 10.0→ANSYS**. That will bring you to the main ANSYS **Utility Menu** as seen here in \$.

All Programs		ANSYS ANSYS Pro ANSYS Wo	oduct Launo orkbench	cher
ANSYS University Introductory Utility Menu				
Eile Select List Plot PlotCtrls WorkPlane Pa	<u>r</u> ameters	<u>M</u> acro M	Me <u>n</u> uCtrls	Help
D 🛎 🖬 🗗 🖨 🖉 📍 📕 📐				
ANSYS Toolbar AN	JSYS Comm	and Promp	t	
SAVE_DB RESUM_DB QUIT POWRGRPH				
ANSYS Main Menu				
Preferences Preprocessor Solution General Postproc TimeHist Postpro ROM Tool Prob Design Radiation Opt Run-Time Stats Session Editor Finish	DES			

Figure 5 Opening ANSYS to the Utility Menu and graphics window

Select job name and analysis type

The various menus below will sometimes get moved to a back (hidden) window. If you think

that has occurred hit the **Raise Hidden** button, 📧. You will always need a job name:

- 1. Utility Menu→File→Change Jobname.
- 2. Change_Jobname, type in the new name, OK (as seen \$).



Figure 1 Setting the new job name

The ANSYS file sizes for real engineering problems get to be quite large, so have a directory dedicated to ANSYS:

- 1. Utility Menu→File→Change Directory.
- Browse for Folder→Change Working Directory, pick your directory (ANSYS_dir in Figure 2), OK.



Figure 2 Establish a directory for the analysis files

To keep up with your analysis studies over time create descriptive titles:

- 1. Utility Menu→File→Change Title.
- 2. Change Title, enter a descriptive title, OK (see Figure 3).



Figure 3 Provide a descriptive title

Element type data

You will need a 3D structural line element, or frame member, that transmits axial forces, axial torsion, and transverse bending moments. Open and add to element types you need:

- 1. Main Menu→Preferences→Preferences for GUI Filtering.
- 2. Check Structural, accept default h-Method, OK.
- 3. Main Menu→Preprocesser→Element Type→Add/Edit/Delete, as in Figure 4.
- 4. In Element Types, seen in Figure 5, pick Add→Library of Element Types.
- 5. Select (Structural) Beam and 3D elastic 4 (that is, beam4), OK.
- 6. In Element Types pick Close.



Figure 4 Declare the selection of a structural analysis

The Beam4 element is 3D and by default is used in the X,Y,Z space and usually requires planning to orientate the local second area moment of inertia axes, IYY and IZZ.



Figure 5 Select space frame element type

Element geometric properties data

Every element type requires one or more real constants, like area or moment of inertia, to describe it. Here, you need several geometric items for each type of beam. You must enter at least one: IYY (although giving just the area makes it a truss member, which can be a useful trick sometimes). Paraphrasing from the ANSYS manual on the BEAM4 element:

"The geometry, node locations, and coordinate systems for this element are shown in Figure 6]. The element is defined by two or three nodes, the cross-sectional area, two area moments of inertia (IZZ and IYY), two thicknesses (TKY and TKZ), an angle of orientation (Θ) about the element x-axis, the torsional moment of inertia (IXX), and the material properties. ... If IXX is not specified or is equal to 0.0, it is assumed equal to the polar moment of inertia (IYY+IZZ). IXX should be positive and is usually less than the polar moment of inertia. The element torsional stiffness decreases with decreasing values of IXX. ... "

The element x-axis is oriented from node I toward node J. For the two-node option, the default (Θ =0°) orientation of the element y-axis is automatically calculated to be parallel to the global X-Y plane. Several orientations are shown in [Figure 6]. For the case where the element is parallel to the global Z axis (or within a 0.01 percent slope of it), the element y axis is oriented parallel to the global Y axis (as shown). For user control of the element orientation about the element x-axis, use the Θ angle (THETA) or the third node option. If both are defined, the third node option takes precedence. The third node (K), if used, defines a plane (with I and J) containing the element x and z axes (as shown)." [End paraphrasing.]

For circular sections IYY = IZZ and you just need to locate the two end nodes to define the local x-axis in space. But in the common case where IYY and IZZ are different you must exercise care in using the third node to define the local z-axis. That is, the local x-z plane coincides with the plane that contains the line (beam) element and the third (orientation) node, K. The local IZZ is relative to that local z-axis and the IYY is defined relative to the local y-axis which is defined by the "right hand rule" from the local x- and z-axes. *Experience shows that this is the most common user error in employing 3D beam elements*.

The selection of the third node occurs later, when the element connectivity is picked in the graphics area, or read fro a file. Sometimes it is useful to have a few nodes just for orientation use, if the needed structure nodes do not do that. If that approach is used they should be completely restrained to avoid any potential problems in the equation solver.

To enter the various real constants to be referenced by the element types:

- 1. Main Menu \rightarrow Preferences \rightarrow Real Constants \rightarrow Add/Edit/Delete.
- 2. Real Constants \rightarrow Add \rightarrow Element Type for Real Constants.
- 3. Choose element type: Type 1 BEAM4, OK, as in Figure 7.
- 4. In Real Constant Set Number 1, for BEAM4 verify Set No. 1.
- 5. Enter 4.0 (m²) for **Cross-sectional area** (AREA); 1.0 (m⁴) for **IZZ**, 0.3 (m⁴) for **IYY**, and 0.3 (m⁴) for **IXX** (see Figure 8 left). The thicknesses (width and depth) of the beam cross-section were not given in the problem statement. Thus, assume

values of 0.4 (m) and 0.1 (m) in the z- and y-directions for **TKZ** and **TKY**, respectively, to help illustrate a non-circular cross-section (like that in Figure 1), **OK**.

- 6. Set 1 appears in Real Constants. Select Add,
- 7. In Real Constant Set Number 2, for BEAM4 verify Set No. 2.
- Reverse the y- and z- values used in step 5. Enter 4.0 (m²) for Cross-sectional area (AREA); 1.0 (m⁴) for IYY, 0.3 (m⁴) for IZZ, and 0.3 (m⁴) for IXX, 0.1 (m) for TKZ and 0.4 (m) for TKY (see Figure 8 right), OK.
- 9. Set 2 appears in Real Constants. Select Close.



Figure 6 Beam4 orientation data from ANSYS manual



Figure 7 Starting the first set of real constants

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▲ Real Constant Set Number	1, for BEAM4	-	Real Constant Set Number 2, for BEAM4	
Element Type Reference No. 1		-	Element Type Reference No. 1	1
Real Constant Set No.	[Þ,	Real Constant Set No.	2
Cross-sectional area AREA	[4	Cross-sectional area AREA	4
Area moment of inertia IZZ		0.3	Area moment of inertia IZZ	1
Area moment of inertia IYY	[1	Area moment of inertia IYY	0.3
Thickness along Z axis TKZ	[0.4	Thickness along Z axis TKZ	0.1
Thickness along Y axis TKY	[0.1	Thickness along Y axis TKY	0.4
Orientation about X axis THETA	[0	Orientation about X axis THETA	
Initial strain ISTRN	[0	Initial strain ISTRN	
Torsional moment of inertia IXX	[0.6	Torsional moment of inertia IXX	0.6
Shear deflection const Z SHEARZ		0	Shear deflection const Z_SHEARZ	
Shear deflection const Y SHEARY		0	Shear deflection const Y_SHEARY	
Rotational frequency SPIN	[0	Rotational frequency SPIN	
Added mass/unit length ADDMA	5	0	Added mass/unit length ADDMAS	
{-	OK		Add Close	

Figure 8 Real constants for beams 1 and 2, and for beam 3, respectively

Note that there are 12 items for each real constants set. The same data can be entered through a text file, as seen at the end of this example, but only 6 items are allowed on each of two lines of input.

One use of the attributes for the ThicKness in the Z (TKZ) and Y (TKY) directions is to plot the approximate shape of each member in its local coordinate system, as seen in Figure . To create such a plot:

- 1. Utility Menu→PlotCtrls→Style→Size and Shape ...
- 2. Size and Shape→ Element SHAPE (ESHAPE) on. Shrink if desired (Figure 9).

Plot <u>C</u> tris <u>W</u> orkPlane Parameters <u>M</u> acro MenuCtris		
Pan Zoom Rotate	Size and Shape	1
View Settings	[/SHRINK] Shrink entities by	10 percent
Numbering	[/ESHAPE] Display of element	🔽 On 🚦
Symbols	shapes based on real	4
Style 🔹 🕨 👌 Style	constant descriptions	3
Font Controls Window Controls Edge Options	SCALE Real constant multiplier	1



Define member material properties:

Here you will use the simplest linear, isotropic, 1D material description. ANSYS has full anisotropic (completely directionally dependent), as well as non-linear material "constitutive laws". Activate the material properties with:

- 1 Main Menu→Preprocessor→Material Props→Material Models.
- 2. Material Model Number 1 appears in **Define Material Model Behavior**.
- 3. Double click on Structural, then Linear, then Elastic, then Isotropic.
- 4. In Linear Isotropic Properties for Material Number 1 enter 1.e6 (kN/m²) for elastic modulus, **EX**, and 0.0 for Poisson's ratio (**PRXY**), as in Figure 10, **OK**.
- 5. Close (X) the Define Material Model Behavior window.

b.

Preprocessor	▲ Define Material Model Behavior	
Element Type	Material Edit Favorite Help	Ę
Real Constants	- Makazial Madala DaGaad	
Material Props	Material Models Derined	Material Models Available
Material Library	Aterial Model Number 1	Favorites
Imperature Units		Structural
Material Models		
Convert Al Px		
\Lambda Linear Isoti	ropic Properties for Material Number 1 🕤	}
Linear Isotropic	: Material Properties for Material Number 1 🛛 👌	
		}
	T1	ζ
Temperatures		[
EV	1.06	· ······

Figure 10 Define the isotropic elastic properties

OK.

Define nodal data:

PRXY

Of course, ANSYS has powerful mesh generation capabilities. However, for beginners or small problems with only a few nodes you can type in the coordinates, or use cursor input via the graphics window, or read them from a file. Use the first approach:

- 1. Main Menu→Preprocessor→Modeling→Create→Nodes→In Active CS.
- 2. In Create Nodes in Active Coordinate System of Figure 11 enter 1 for Node number, X = 0., Y = 5., Z = 5 (m), Apply, enter 2 for Node number, X = 5., Y = 5., Z = 5., Apply, enter 3 for Node number, X = 5., Y = 5., Z = 0., Apply, enter 4 for Node number, X = 5., Y = 0., Z = 0., Apply.
- 3. You can also add a non-structural node for beam orientation purposes. If so, enter 5 for **Node number**, **X** = 5., **Y** = 0., **Z** = 5., **Apply, OK.**
- 4. If you make a mistake you can return and correct it in the above window, or delete them all with **Preprocessor→Modeling→Delete**.

Preprocessor	Create Nodes in Active Coordinate System					
Element Type Deal Constants	[N] Create Nodes in Active Coordinate System	[N] Create Nodes in Active Coordinate System				
Material Props	NODE Node number					
Sections						
Modeling	X,Y,Z Location in active CS	0 5 5				
🗆 Create	THXY,THYZ,THZX					
Keypoints	Rotation angles (degrees)					
🗄 Lines						
H Areas						
➢ On Working Plane						
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2	3					
7	المستقلة المستعدية المستعد المس					
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Figure 11 Build the list of nodal coordinates

Now, plot the nodal values input (node 1 may be hidden behind the axis symbol):

- 1. Utility Menu→PlotCtrls→Numbering.
- 2. In **Plot Numbering Controls** check **node numbers** and select **element numbers** in the pull down menu, **OK**.
- 4. To get the reverse video white background of that figure use PlotCtrls→Style→ Color→Reverse Video.

# Define element attributes

Next you have to associate each of the elements with your previous material numbers and real constant sets. Plan ahead and input those elements of the same type in sequence:

- 1. Main Menu→Preprocessor→Modeling→Create→Elements→Elem Attributes.
- 2. In **Element Attributes** select defaults (type number = 1, material number = 1, real constant set = 1), **OK**.
- 3. Main Menu→Preprocessor→Modeling→Create→Elements→ Auto numbered→Thru Nodes, as in Figure 13.
- 4. In Elements from nodes verify that Pick is checked on.
- 5. In the graphics window, seen in Figure 14, define the first element by picking **node 1** (a square symbol appears) then **node 2**, then orientation **node 3**, **OK**, and the next element number (1) appears in the graphics area.
- 6. Repeat this process for the last member sharing the current real constants set: **Thru Nodes**, pick **nodes 2, 3**, and **1** for element 2 (bottom of Figure 14).



Figure 12 Confirm the beam attributes for the first two members



Figure 13 Prepare to graphically pick the element connectivity

For the next (and last) element you need to change the number of its real constants data set from 1 to 2. If those data were still correct but the next element was made of a different material then a similar change would be required at this point. To accomplish the change, seen in Figure 15, use:

1. Preprocessor→Modeling→Create→Elements→Elem Attributes.

2. In **Element Attributes** change the **Real constants set number** from 1 to 2.

If you are using text file input rather than the GUI then a similar change (REAL,2) is needed.

Now you can continue with picking the nodes for the final member in the mesh.

- 1. Repeat the previous connectivity definition process.
- 2. Thru Nodes, pick nodes 3, 4, and 2 for element 3, as illustrated in Figure 16.



Figure 14 Define two elements for first real constants set

∧Element Attributes		
Define attributes for elements		
[TYPE] Element type number	1	BEAM4
[MAT] Material number		1
[REAL] Real constant set number		2

Figure 15 Set new real constants set for next element(s)



Figure 16 Final element pick for the space frame mesh

#### Check mesh data

It is wise to check such element input by plotting the nodes and elements, and/or the element's real constant set numbers via:

- 1. Utility Menu→PlotCtrls→Numbers ...
- 2. Plot Numbering Controls → Elem /Attrib numbering pick real const num in Figure 17 (left) to create the associated display (right).



Figure 17 Displaying the real constant set numbers

Likewise, you can view a list of the data with:

- 1. Utility Menu→List→Elements→Nodes+Attr+RealConst.
- 2. When the ELIST (ELement LIST) window appears check those data and close it.

List	Plot	Plot <u>⊂</u> trls	<u>W</u> orkPlane	Parameters }
F	iles tatus	+		
K Li A	eypoinl ines ireas	: •	ŧРН	
V N	olumes Iodes	•	8 -	
E	lement:	s 🕨	Nodes +	Attributes

ELIST Comma	and					
e						
IST ALL SEI	LECTED ELEMEN	TS. (LI	ST NODES>			
FLEM MOT	TVP REL ESV	SEC	NODES			
EBEN INT	III ALL LOI	OLO	HOPLO			
1 1	1 1 0	1	1 2	3		
AREA	IZZ	IYY		TKZ	TKY	THET
ISTR	IXX	SHRZ		SHRY	SPIN	ADMS
4.00000	0.300000	1.00	กดด ด	.400000	0.100000	0.0000
0.00000	0.600000	0.00	กดด	0.00000	0.00000	0.0000
2 1	1 1 0	1	2 3	1		
4.00000	0.300000	1.00	אמא א	.400000	0.100000	0.0000
0.00000	0.600000	0.00	บดด	0.00000	0.00000	0.0000
3 1	1 2 0	1	3 4	2		
AREA	IZZ	IYY		TKZ	TKY	THET
ISTR	IXX	SHRZ	la a la constante de la consta	SHRY	SPIN	ADMS
4.00000	1.00000	0.300	000 0	100000	0.400000	0.0000
0.00000	0.600000	0.00	000	0.00000	0.0000	0.0000

Figure 18 Listing the element attributes for checking

### Apply displacement restraints

The displacement restraints must be applied to reflect the physical support (often the most unclear part of an analysis) as well as eliminating all the "*rigid body motions*" (RBM). Here there are three translational RBM plus three rotational ones. Apply the system nodal displacement restraints via:

- Main Menu→Preprocessor→Loads→Define Loads→ Apply→Structural→Displacement→On Nodes (see Figure 19).
- 2. In Apply U, Rot on Nodes verify that Pick and Single are checked on.
- 3. In the graphics window, Figure 20, select top **node 1** and bottom **node 4** (for complete restraints), **OK**. After the panel changes form, highlight the fixed option for all of the degrees of freedom components (**All DOF**) as the **DOF to be constrained**
- 4. Under Constant value enter 0, OK.

Note that these restraint operations are shown in the graphics window as triangles pointing in the direction of restraint, at each restrained node. To list the current restraints:

#### 1. Utility Menu→List→Loads→DOF Constraints→On All Nodes.

2. When the (**D**isplacement **LIST**) **DLIST window** appears check those data and close it Review the graphical restraint symbols in Figure 20 to verify the choice you picked from the list of available displacement restraints.



Figure 19 Preparing to graphically assign displacement restraints



Figure 20 Picking nodes to restrain all displacements and rotations

# Apply nodal loads

This space frame is intended to carry a vertical load of -100 (kN) at node 2. Specify that y-component force with:

- 1. Main Menu→Preprocessor→Loads→Define Loads→Apply→Structural→Force/Moment→On Nodes.
- 2. In Apply F/M on Nodes verify that Pick and Single are on, and then pick node 2, OK.
- 3. When the window changes form (lower part of Figure 21, pick a vertical force (**FY**) for the **Direction** of force and -100 (kN) for the **Constant value**, **OK**.
- 4. Then the arrows representing the load should appear as in Figure 22.



Figure 21 Graphically picking loaded nodes (lower left) and assigning values

Note that these load operations are shown in the graphics window as arrows pointing in the direction of load, at each restrained node. If your plot does not show everything you have defined also try **PlotCtrls**->**Symbols**...->**Symbols** and check that the items you want are turned on.



Figure 22 Loaded node's force arrow(s) appear

You may also wish to list the current loads:

- 1. Utility Menu→List→Loads→Forces→On All Nodes.
- 2. When the Force LIST (FLIST) window appears check those data and close it.

#### Saves and restarts

At this point you may want to save your data and restart the actual analysis later. If so:

- 1. Utility Menu→File→Save as Jobname.db.
- 2. Upon return open ANSYS and enter **Utility Menu→File→Resume from...** to get the list of your ANSYS database (db) files so you can select the one you want.

### Solve for displacements and secondary variables

To use the current (and only) load system (LS) enter:

- 1. Main Menu-Solution-Solve-Current LS, review the listed summary, OK.
- 2. When the solution of the simultaneous equations is complete you will be alerted that the solution is done.



Figure 23 Solving displacements for the current load set

# **Post-processing**

#### Displacements

I

t is always wise to visually check the computed displacements:

- 1. Main Menu→General Postproc→Plot Results→Deformed Shape.
- 2. In **Plot Deformed Shape** pick the combined **deflected and undeformed** option for the Items to be plotted (KUND), **OK**. Check the plot in the graphics window, such as the one seen in Figure 24.
- 3. Animate the computed deflections with: Utility Menu→PlotCrtls→Animate→ Deformed Shape.
- 4. Pick Def+undeformed in Animate Deformed Shape, OK.
- 5. If desired, employ the Animation Controller that appears, or simply pick Stop, **Close**.





Figure 24 The deformed and undeformed space frame

To create a hardcopy (you may need to try various background colors):

- Utility Menu→PlotCrtls→Hard Copy→Printer (or →File), select your printer name, Print.
- 2. To get the reverse video white background of Figure 24 use **PlotCtrls→Style→ Color→Reverse Video**.

Since displacements and (infinitesimal) rotations are vector quantities it is wise to plot them in that mode as a visual check of the response of the system. To do that:

- 1. General Postproc→Plot Results→Vector Plot→Predefined.
- 2. In Vector Plot of Predefined Vectors select DOF solution, Translation U, Vector mode, and element nodes.

The resulting color plot will display the vectors with scaled lengths and with a color matching the color bar scale, as seen in Figure 25.



Figure 25 Nodal displacement vector plot

To see a (potentially long) list of displacement results:

- 1. Preferences→General Postproc→List Results→Nodal Solution.
- 2. In List Nodal Solution→Nodal Solutions→DOF Solution→Displacement vector sum, OK.
- 3. Examine the results in the **PRNSOL** (**PR**int **N**odal **SOL**ution) **Command window** of Figure 26 and close it.

4. Likewise, to see the nodal rotations, also in Figure 26,use List Nodal Solution→ Nodal Solutions→DOF Solution→Rotation vector sum, OK.



RNSOL Command
File
PRINT U NODAL SOLUTION PER NODE
***** POST1 NODAL DEGREE OF FREEDOM LISTING *****
LOAD STEP= 1 SUBSTEP= 1 TIME= 1.0000 LOAD CASE= 0
THE FOLLOWING DEGREE OF FREEDOM RESULTS ARE IN THE GLOBAL COORDINATE SYSTEM
NODE UX UY UZ USUM 1 0.0000 0.0000 0.0000 0.0000 2 -0.30390E-05-0.59965E-02 0.87687E-03 0.60603E-02 3 0.95706E-03-0.45359E-04 0.91129E-03 0.13223E-02 4 0.0000 0.0000 0.0000 0.0000
MAXIMUM ABSOLUTE VALUES NODE 3 2 3 2 VALUE 0.95706E-03-0.59965E-02 0.91129E-03 0.60603E-02

```
MPRNSOL Command
File
 PRINT ROT NODAL SOLUTION PER NODE
  ***** POST1 NODAL DEGREE OF FREEDOM LISTING *****
  LOAD STEP=
                      SUBSTEP=
                   1
                                     1
            1.0000
                          LOAD CASE=
                                         Ø
   TIME =
  THE FOLLOWING DEGREE OF FREEDOM RESULTS ARE IN THE GLOBAL COORDINATE SYSTEM
    NODE
               ROTX
                             ROTY
                                           ROTZ
                                                        RSUM
            0.0000
                         0.0000
                                       0.0000
                                                     0.0000
       1
           0.11295E-02-0.23600E-03-0.15137E-02 0.19033E-02
0.74700E-03-0.15816E-03-0.37269E-03 0.84966E-03
       2
       3
                         0.0000
            0.0000
                                       0.0000
                                                     0.0000
       4
 MAXIMUM ABSOLUTE VALUES
 NODE
                              2
          0.11295E-02-0.23600E-03-0.15137E-02 0.19033E-02
 VALUE
```



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#### Displacement solution validation

The generalized nodal displacements shown in Figure 26 agree with the (four significant figure) results in [1]. Therefore, you can conclude that you have followed the correct steps for the analysis of a space frame. Next you can review the system reactions, the member reactions and member stresses.

#### System reaction forces

If the solver does not fail then your reactions will be equal and opposite of your resultant forces and moments. Therefore they let you check the level of loads actually applied, versus what you intended to apply. That is helpful especially for pressure loads. Check the system reactions with:

- 1. Main Menu→General Postproc→List Results→Reaction Solution.
- 2. In List Reaction Solution pick All items for Item to be listed, OK.
- 3. Review the Print Reaction SOLution (PRRSOL) Command window of Figure 27.

Those system reactions also agree with [1] to four significant figures.



APRRSOL Command
File
PRINT REACTION SOLUTIONS PER NODE
***** POST1 TOTAL REACTION SOLUTION LISTING *****
LOAD STEP= 1 SUBSTEP= 1 TIME= 1.0000 LOAD CASE= 0
THE FOLLOWING X,Y,Z SOLUTIONS ARE IN THE GLOBAL COORDINATE SYSTEM
NODE         FX         FY         FZ         MX         MY         MZ           1         2.4312         63.713         -27.539         -67.769         116.05         250.10           4         -2.4312         36.287         27.539         24.027         9.4899         80.616
TOTAL VALUES VALUE -0.19096E-13 100.00 -0.11369E-12 -43.742 125.54 330.72

Figure 27 Examine the system reaction forces and moments

#### Member end forces and moments

The element (member) end forces and moments, also known as the member reactions, in the space frame resulting from the computed displacements can also be recovered and listed:

- 1. Main Menu $\rightarrow$ General Postproc $\rightarrow$ List Results $\rightarrow$ Element Solution.
- 2. Get the member axial forces, in global components, from List Element Solution→Element Solution→All Available Force Items, OK.

3. Review the **PRESOL** (**PR**int **EL**ement **SOL**ution) **Command** display of Figure 28.

Those element level reactions also agree with [1] to four significant figures.

<b>N</b> PRESOL (	Command					
File						4
		ALUTION DED				
PRINI FU	KC ELEMENT :	SOLUTION PER	ELEMENI			1
***** PO	ST1 ELEMENT	NODE TOTAL	FORCE LISTIN	G <del>****</del>		
LOAD ST TIME=	EP= 1 % 1.0000	SUBSTEP= LOAD CAS	1 E= 0			ĺ
THE FOL	LOWING X,Y,2	Z FORCES ARE	IN GLOBAL C	OORDINATES		5
ELEM=	1 FX	FY	FZ	MX	MY	MZ 3
1	-2.4312	-63.713	27.539	67.769	-116.05	-250.10
2	2.4312	63.713	-27.539	-67.769	-21.646	-68.461
ELEM=	2 FX	FY	FZ	MX	MY	MZ
2	-2.4312	36.287	27.539	67.769	21.646	68.461
3	2.4312	-36.287	-27.539	113.67	-9.4899	-68.461
ELEM=	3 FX	FY	FZ	MX	MV	MZ
3	-2.4312	36.287	27.539	-113.67	9.4899	68.461
4	2.4312	-36.287	-27.539	-24.027	-9.4899	-80.616
						1

Figure 28 Individual member end reactions

#### Member stress state

Reference [1] does not give the member stresses. ANSYS does give them and the strains also. Here, the stress data, other than the neutral axis stress SDIR, are not meaningful because the Y- and Z-thickness were not known but were chosen to allow visual checks, as seen in Figure . When thickness values consistent with the area, IYY, and IZZ are given then the reported stresses are the bending stresses superimposed on the axial stresses.



Figure 29 Definition of the BEAM4 stresses

Page 23 of 26 Copyright J.E. Akin. All rights reserved.

🖃 General Postproc	
📷 Data & File Opts	
Results Summary	
Read Results	
표 Failure Criteria	
Plot Results	
List Results	List Element Solution
📰 Detailed Summary	□ Item to be listed
📰 Iteration Summry	
📰 Percent Error	🛃 Favorites
Sorted Listing	Element Solution
Nodal Solution	Change 1
Element Solution	Stress
	Line Element Results

Figure 30 Selecting line element stress results

RESOL Command					
File					
BRINE FIEM FIEMENT COLUTION DED FIEMENT					
PRINI ELEM ELEMENI SOLUTION PER ELEMENI					
***** POST1 ELEMENT SOLUTION LISTING *****					
LOAD STEP 1 SUBSTEP= 1 TIME= 1.0000 LOAD CASE= 0					
EL=	1 NODES =	1	2	3 MAT= 1	4
BEAM4 TEMP =	0.00 0.00	0.00	0.00	0.00 0.00	0.00 0.00
LOCATION	SDIR	SBYT	SBYB	SBZT	SBZB
1 (1)	-0.60780	-41.684	41.68	4 23.209	-23.209
	-0.60780 CMOY	11.410 CMIN	-11.41	0 -4.3292	4.3292
1 (I)	64.286	-65.501			1
2 (J)	15.131	-16.347			1
II		1- 11-01.00	Care Same		بر و مربع بری از مربع میشون میشود. بر و مربع بری از مربع میشون میشود می مربع ا
EL=	2 NODES =	2	3	1 MAT= 1	
BEAM4	0 00 0 00	0 00	0 00	0 00 0 00	0 00 0 00
LOCATION	0.00 0.00 SDIR	SBAL SBAL	0.00 SBAB	0.00 0.00 SBZT	9.00 0.00 SBZB
1 (I)	-6.8846	11.295	-11.29	-4.3292	4.3292
2 (J)	-6.8846	-18.944	18.9	44 -1.8980	1.8980
LOCATION	SMAX	SMIN			2
1 (1)	8.7393	-22.509			
2 (J)	13.958	-27.727	ENT. DAL		STREET DOTT ON A STREET
NATE AND AND		5-1-1-1-1-1-	and the second second	Contraction of the second second	a a carrente de la c
EL= BEOMA	3 NODE2=	3	4	2 MHI= 1	
TEMP =	0.00 0.00	0.00	0.00	0.00 0.00	0.00 0.00 ¹
LOCATION	SDIR	SBYT	SBYB	SBZT	SBZB
1 (I)	-9.0717	-13.692	13.69	2 -18.944	18.944
2 (J)	-9.0717	-16.123	16.12	3 4.0044	-4.0044 刘
LOCATION	SMAX	SMIN			1
	23.505	-41.708			1
havenan					and the second second second

* S = Stress, DIR = DIRect (axial), BYT, BYB = Bending Y-axis Top, Bottom, BZT, BZB = Bending Z-axis Top, Bottom



# Exiting ANSYS

You can save the results to your data base, as described above, and close with **File→Exit**. To begin a different analysis, instead of picking Exit, use **File→Clear & Start New ..., OK**.

### Closure

For this class of loading this three element model will be close to giving exact results for the generalized nodal displacements. The beam deflections are cubic in space but most finite element codes execute a linear plot between the two end nodes, as in Figure 24. If you wanted to see the more realistic deformed shape the simplest thing to do is to divide each of the elements into more elements, say 10 each. ANSYS can easily do that for you.

The job data for almost any analysis can be entered by a text file. The data for this example are in Figure 32. One thing you should note about this alternate input approach is that the data for the real constants sets will only accept six entries per line. Therefore, keyword *r* (for real) must be followed with the keyword *rmore* in order to supply the IXX term here. Lines beginning with a slash (/) are ANSYS commands. Comments are preceded with an exclamation symbol (!) and extend to the end of the line.

To read such a data text file into ANSYS begin in the **Utility Menu**:

- 1. File→Read Input From ...
- 2. Pick the **Directory**, **Filename**, **OK**.

Then you can use the GUI to pick up your study at the stage where you chose to terminate the batch text file. The entire analysis process can be controlled from a batch file. Experienced users often find this feature to be a time saver.

### Reference:

I.M. Smith, D.V. Griffiths, Programming the Finite Element Method, 3rd Ed, Chichester, John Wiley, 1998

```
/FILNAM, S G SF 44
/title, Smith Griffths space frame Ex. 4.4b
/prep7
              General Summary
! Smith Griffths space frame Ex. 4.4b page137-141
! Units: m, kN, m<sup>A</sup>2, m<sup>A</sup>4, kN/m<sup>A</sup>2
! AREA = 4, IX = 0.3, IY = 1, IZ = 0.3, P = 100
! ELASTIC = 1.e6, SHEAR = 1.e6
•
  #1 Fixed
                   *3
                          Node 2 displacements:
           Р
ŗ.
                 71
                          -3.039E-6 -5.997E-3 8.769E-4
   ١.
ļ
          :
               1
                          Node 2 rotations:
     ١.
                    - I
    (1) : /(2) |(3) 1.130E-3 -2.360E-4 -1.514E-3
ļ
ļ
         \ v/
                          Node 1 force reactions:
                    1
L
           *2
                          2.431E+0, 6.371E+1, -2.754E+1
                    ļ.
                    #4 Fixed
          5, 5 ! begin node definition
n, 1, O,
n, 2, 5, 5,
                5
         5,
n, 3, 5,
                0
n, 4, 5,
         0,
                0
et,1,beam4
                1
                     first element type
mp,ex,1,1e6
                . I
                     first material type
mp,prxy,1,0.0
     first real constant set (elements 1 & 2)
I.
     AREA, IZZ, IYY, TKZ, TKY, THETA
                                      (6 MAX !)
1
r, 1, 4., 0.3, 1.0, 0.4, 0.1,
                                0
    ISTRN, IXX, SHEARZ, SHEARY, SPIN, DDMAS
!
                      Ο,
rmore, 0, 0.6,
                              Ο,
                                   Ο,
                                           0
     second real constant set (elements 3)
     AREA, IZZ, IYY, TKZ, TKY, THETA
                                       (6 MAX !)
1
r, 2, 4., 1.0, 0.3, 0.1, 0.4, 0
     ISTRN, IXX, SHEARZ, SHEARY, SPIN, ADDMAS
1
                                   Ο,
rmore, 0, 0.6,
                             Ο,
                      Ο,
                                            0
     Define elements of et=1, mt=1. First 2 nodes set
I
.
     local beam x-axis, third node sets local xz-plane
REAL,1
                 ! turn on real property set #1
en, 1, 1, 2, 3 ! elem i, j, k
en, 2, 2, 3, 1
REAL,2
                 ! activate real property set #2
en, 3, 3, 4, 2
d, 1, all, 0.
                 ! displacement restraints
d, 4, all, 0.
f, 2, fy, -100. ! force at joint
finish ! batch input, go to GUI
```

Figure 32 Text file to describe this example mesh, restraints, and loads